# An Experimental Study of the Formation and Decomposition of Nitric Oxide 

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TTHE PRIMARY reaction involved in the formation and decomposition of nitric oxide is described by Equation 1:

$$
\begin{equation*}
2 \mathrm{NO} \rightleftharpoons \mathrm{~N}_{2}+\mathrm{O}_{2} \tag{1}
\end{equation*}
$$

Nerst (8) and Jellinek (4) evaluated the equilibrium of the reaction described by Equation 1 and expressed results in terms of an equation constant defined as follows:

$$
\begin{equation*}
\frac{(\mathrm{NO})_{e}^{2}}{\left(\mathrm{~N}_{2}\right)_{e}\left(\mathrm{O}_{2}\right)_{e}}=K=\frac{k_{2}}{k_{1}} \tag{2}
\end{equation*}
$$

Values of this conatant were tabulated by Rossini (11). Penner (10) studied many aspects of the formation and decomposition of the oxides of nitrogen, and Vetter (12) and Wise (14) considered the influence of oxygen as a catalyst in accelerating the decomposition of nitric oxide.

Figure 1 presents semiquantitatively the effects of temperature upon the equilibrium mole fraction of nitric oxide for a pressure of 1000 p.s.i. In carrying out these calculations, it was assumed that there were equal mole fractions of nitrogen and oxygen initially. Only the reactions pertaining to the formation of nitric oxide from nitrogen and oxygen and the formation of monatomic nitrogen and oxygen from diatomic nitrogen and oxygen were considered. Account was taken of the fact that the formation of monatomic oxygen tends to increase the mole fraction of nitric oxide at the higher pressures. The data compiled by Lewis and von Elbe (5) were employed.
Figure 1 also shows the mole fraction of nitric oxide which remains after cooling the system at different rates of temperature change. This latter information was obtained following the general methods suggested by Altman and


Figure 1. Effect of rate of cooling on residual composition

Penner (1). Kinetics data of the form suggested by Gilbert and Daniels (3) were employed to establish the curves. The mole fraction of nitric oxide at equilibrium increases in a regular fashion with an increase in temperature, and it is possible to take advantage of the increased mole fraction of nitric oxide, if a high rate of cooling is utilized. A rate of cooling of at least $10^{8} \circ \mathrm{R}$. per second is required to obtain more than 0.04 mole fraction nitric oxide.

## EQUIPMENT

For the present investigation a ballistic piston apparatus (7) was used. Figure 2 shows, schematically, the heavywalled cylinder, $A$, within which the piston, $B$, moved. The sample was confined in the space, $C$, beneath the piston. Compressed gas contained in the space, $D$, drove the piston following its release by a shear pin, $E$. Electrical contacts, $F$, were used to determine the position of the piston as a function of time. The composition of the initial gas sample, the thermal transfer from the sample, and the composition of the gas sample after compression also were determined. Pressure ratios as high as 100,000 to 1 were obtained, and by appropriate choice of diluent gases it was possible to reach


Figure 2. Schematic diagram of apparatus
energies equivalent to apparent temperatures of $100,000^{\circ} \mathrm{R}$.
Figure 3 illustrates the general nature of the temperaturetime relationship during typical compression cycles involving the heavy and light pistons used. The data show that the temperature course of the compression and expansion above $3000^{\circ} \mathrm{R}$. was limited to approximately 4 milliseconds for the light piston and to approximately 7 milliseconds for the heavy piston. The temperature data of Figure 3 were obtained by calculations assuming isentropic compression and expansion and no chemical reaction,


Figure 3. Temperature-time relations for two pistons for a composition of 0.4 mole fraction nitrogen, 0.2 mole fraction oxygen, and 0.4 mole fraction helium.
whereas the time data were based on actual measurements made on the downstroke of the piston during each test. Throughout this article, the term temperature used in connection with the state of the sample refers to apparent temperature.

## METHOD OF ANALYSIS

Measurements of the temperature and pressure of each component of the initial gas sample were made during introduction to the sample chamber. For most of the tests, the composition of the initial sample was established from these data. In the 13 earliest tests, mass spectrographic analyses were made on samples of the initial gas which were withdrawn following introduction to the sample chamber.

After compression, samples of the product were withdrawn and stored in sealed glass bulbs. The product compositions for the first 19 tests were determined by mass spectrographic analysis. This technique was unsatisfactory for the determination of the oxides of nitrogen, however, and it was replaced by a chemical procedure involving titrimetric methods (2). This procedure involved introduction of the gas sample to an evacuated flask containing a neutral hydrogen peroxide solution. The oxides of nitrogen were converted to nitric acid which was titrated with standard alkali. Experience with these analytical techniques indicated that the mole fraction of oxides of nitrogen could be determined within 0.001 . Results which showed a larger deviation were included only in a few instances to show a trend in a region where no other data were at hand.

## EXPERIMENTAL RESULTS

The experimental conditions for each of 65 tests upon the nitrogen-oxygen system are set forth in Table I. The maximum apparent temperatures given in Table I were calculated ( 6,7 ) from measurements of the closest approach of the piston to the bottom of the cylinder, assuming isentropic compression and no chemical reaction. The maximum pressure ratio given in the table is defined as the ratio of the maximum pressure attained by the sample to the initial pressure of the sample. The conditions set forth for minimum volume are subject to uncertainty, but they present the trends in the results.
the relatue thermal transport from the sample (Table I) is expressed as the ratio between the thermal transfer from
the sample and the maximum apparent energy transferred to the sample by compression. These data indicate a trend toward an increase in the fractional energy loss as the apparent temperature increases. The entire thermal transfer was established by measuring the radiant and convective transfer to a small area at the bottom of the compression chamber and applying this value per unit area to the face of the piston and bottom of the compression chamber. The varying area associated with the cylinder walls was neglected. The thermal transfer to the piston face and the bottom of the compression chamber may amount, in the case of the higher temperatures, to as much as $13 \%$ of the total energy added to the sample.

If the shear along the walls is neglected, the temperature distribution within the sample is nonuniform for a distance amounting to approximately $5 \%$ of the radius of the cylinder (13). Beyond this area near the walls, within which conduction to the walls exerts an influence, the sample is of a relatively uniform energy content with respect to the spatial coordinates of the sample.

Table II presents the initial and final compositions for the 65 tests. Helium was employed as an initial component in order that the fugacity-temperature relationship of the other components could be varied independently. From a kinetics standpoint, the helium undoubtedly influenced the rate of reaction because it acts as a major diluent. The final compositions are expressed in terms of mole fractions of nitric oxide, although at room temperature a substantial part of the nitric oxide reacted with the remaining oxygen to form nitrogen tetroxide. Table II also includes a residual nitric oxide ratio which is the ratio of the square of the mole fraction of nitric oxide to the product of the mole fraction of nitrogen and oxygen remaining in the product. The values were based on the assumption that all the residual oxides of nitrogen were nitric oxide. This ratio provides a convenient means of comparing the behavior of samples of differing compositions which underwent the same environmental compression cycle.

The influence of the maximum apparent temperature upon the residual nitric oxide ratio is shown in Figures 4 to 6 for compositions containing 0.4 mole fraction nitrogen, 0.2 mole fraction oxygen, and 0.4 mole fraction helium. The data of Figure 4, limited to an initial sample pressure of approximately 1.16 p.s.i., include information for both the light and heavy piston. A markedly higher nitric oxide ratio for the same maximum apparent temperature was found for the tests in which the light piston was used. In Figure 5, the initial sample pressure of approximately 2.31 p.s.i. is nearly twice the pressure in Figure 4. At this higher pressure, the smaller difference in rates of cooling between the tests made with the light piston and those made with the heavy piston produced a smaller difference in the residual amount of nitric oxide than was found at the lower pressure associated with the data depicted in Figure 4. Figure 6 presents data for an initial sample pressure of approximately 4.62 p.s.i. which shows much less, if any, difference between the behavior with the light and the heavy piston.

Figure 7 shows that variation in the nitrogen-oxygen ratio has a relatively small influence upon the relationship of the residual nitric oxide ratio to the maximum apparent temperature. Measurements upon the two compositions illustrated were made with a piston which weighed 3.29 pounds. The straight lines were fitted to the points by a least squares method (9). The residual nitric oxide ratio is roughly 0.0005 larger for the mixture containing the smallerquantity of oxygen. In each case, the original sample contained 0.8 mole fraction helium. These limited data illustrate the measurable influence of oxygen upon the decomposition of nitric oxide, as reported by Vetter (12) and Wise (14).

Approach to chemical equilibrium in the ballistic piston

Table 1. Experimental Conditions for Nitrogen-Oxygen System Tests

|  |  | Initial Conditions ${ }^{\text {c }}$ |  |  |  | Conditions at Minimum Volume ${ }^{b}$ |  |  |  | RelativeThermalTransport$\underline{Q} \Delta \Delta \underline{E}$ | Rate of Temp. - F./sec. |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Test | Piston Weight, Lb. | $\begin{gathered} \hline \text { Driving } \\ \text { air } \\ \text { pressure, } \\ \text { p.s.i. } \end{gathered}$ | $\begin{gathered} \text { Initial } \\ \text { sample } \\ \text { pressure, } \\ \text { p.s.i. } \end{gathered}$ | Sample volume, cu. ft. | Piston approach, in. | Volumetric compression ratio | Maximum apparent temp., ${ }^{\circ} \mathrm{R}$. | Maximum apparent pressure, p.s.i | Maximum pressure ratio |  |  |
| 133 | 2.44 | 684.9 | 3.100 | 0.4282 | 0.1583 | 55.30 | 5100 | 22300 | 7200 |  |  |
| 136 | 2.44 | 570.5 | 2.098 | 0.4282 | 0.1414 | 61.93 | 7800 | 24800 | 11800 |  |  |
| 391 | 3.28 | 343.6 | 2.872 | 0.4282 | 0.9830 | 8.894 | 7000 | 4000 | 1400 | 0.05115 | 0.868 |
| 392 | 3.28 | 500.0 | 2.881 | 0.4282 | 0.4709 | 18.57 | 10300 | 12700 | 4400 | 0.08948 | 2.093 |
| 394 | 3.28 | 292.5 | 2.869 | 0.4282 | 1.2020 | 7.274 | 6300 | 2900 | 1000 | 0.04981 | 0.782 |
| 395 | 3.28 | 238.1 | 2.883 | 0.4282 | 1.9601 | 4.483 | 4900 | 1400 | 500 | 0.03276 |  |
| 397 | 3.28 | 406.1 | 2.880 | 0.4283 | 0.6879 | 12.70 | 8500 | 7200 | 2500 |  | 1.453 |
| 399 | 3.28 | 344.7 | 2.888 | 0.4283 | 0.9691 | 5.716 | 7100 | 4300 | 1500 | 0.07909 | 0.991 |
| 401 | 3.28 | 400.9 | 2.884 | 0.4283 | 0.7021 | 12.55 | 8400 | 6900 | 2400 | 0.07241 | 1.420 |
| 402 | 3.28 | 473.4 | 2.879 | 0.4283 | 0.5661 | 15.61 | 9400 | 9500 | 3300 | 0.05063 | 1.808 |
| 403 | 3.28 | 311.9 | 2.893 | 0.4283 | 1.0969 | 7.966 | 6600 | 3400 | 1200 |  | 0.803 |
| 405 | 3.28 | 270.0 | 2.880 | 0.4283 | 1.4177 | 6.162 | 5800 | 2300 | 800 | 0.07236 | 0.585 |
| 406 | 3.28 | 240.2 | 2.887 | 0.4283 | 1.8452 | 4.739 | 5100 | 1400 | 500 | 0.05134 | 0.351 |
| 407 | 3.28 | 301.7 | 2.888 | 0.4283 | 1.3994 | 6.243 | 5800 | 2300 | 800 | 0.08449 | 0.651 |
| 408 | 3.28 | 344.7 | 2.873 | 0.4283 | 1.1374 | 7.676 | 6500 | 3200 | 1100 | 0.07322 | 0.897 |
| 409 | 3.28 | 400.8 | 2.887 | 0.4283 | 0.6949 | 12.58 | 8400 | 6900 | 2400 | 0.10140 | 1.321 |
| 154 | 3.29 | 522.5 | 2.309 | 0.4282 | 0.2223 | 39,36 | 6900 | 14800 | 6400 |  | 1.418 |
| 162 | 3.29 | 607.1 | 2.307 | 0.4282 | 0.2224 | 39.35 | 6900 | 14700 | 6400 |  | 2.184 |
| 163 | 3.29 | 715.4 | 4.612 | 0.4282 | 0.6635 | 13.17 | 4600 | 6500 | 1400 |  | 0.937 |
| 178 | 3.29 | 1083.0 | 4.633 | 0.4282 | 0.2025 | 43.18 | 7400 | 38900 | 8400 | 0.07685 | 2.551 |
| 179 | 3.29 | 275.4 | 1.061 | 0.4282 | 0.2048 | 42.70 | 7000 | 7200 | 6800 | 0.07232 | 1.254 |
| 184 | 3.29 | 240.6 | 1.191 | 0.4282 | 0.5485 | 15.93 | 4900 | 2100 | 1800 |  |  |
| 185 | 3.29 | 216.3 | 1.160 | 0.4282 | 0.3525 | 24.81 | 5800 | 3800 | 3300 | ... | 0.768 |
| 310 | 3.29 | 317.1 | 2.878 | 0.4282 | $1.3900^{\text {d }}$ | 6.281 | $5900{ }^{\text {d }}$ | $2400^{\text {d }}$ | $850{ }^{\text {d }}$ |  |  |
| 313 | 3.29 | 460.1 | 2.880 | 0.4282 | $1.2080^{\text {d }}$ | 7.228 | $6300^{\text {d }}$ | $3000{ }^{\text {d }}$ | $1000{ }^{\text {d }}$ |  |  |
| 315 | 3.29 | 600.1 | 2.887 | 0.4282 | 0.5468 | 15.99 | 9600 | 10100 | 3500 | 0.13290 | 1.217 |
| 317 | 3.29 | 317.0 | 2.869 | 0.4282 | $1.5730^{\text {d }}$ | 5.550 | $5500^{\text {d }}$ | $2000^{\text {d }}$ | $690^{\text {d }}$ |  |  |
| 319 | 3.29 | 460.1 | 2.878 | 0.4282 | 1.1900 | 7.338 | 6400 | 3200 | 1100 |  |  |
| 321 | 3.29 | 600.1 | 2.885 | 0.4282 | 0.5354 | 16.33 | 9700 | 10400 | 3600 | 0.07063 | 0.658 |
| 146 | 3.38 | 555.1 | 2.115 | 0.4282 | 0.1333 | 6.547 | 8000 | 13100 | 27700 | ... | ... |
| 149 | 3.38 | 207.9 | 1.160 | 0.4282 |  |  |  |  |  | $\ldots$ |  |
| 150 | 3.38 | 437.8 | 2.310 | 0.4282 | 0.3423 | 25.54 | 5900 | 8300 | 3600 |  | 1.444 |
| 152 | 3.38 | 897.3 | 4.616 | 0.4282 | 0.2987 | 29.27 | 6300 | 20800 | 4500 | $\ldots$ | 1.925 |
| 158 | 3.38 | 240.5 | 1.148 | 0.4282 | 0.4505 | 19.39 |  |  |  |  |  |
| 36 | 30.94 | 999.1 | 2.963 | 0.4070 | 0.1104 | 75.14 | 5100 | 25000 | 8400 | $\ldots$ | ... |
| 38 | 30.96 | 1004.0 | 2.297 | 0.4071 |  |  |  |  |  | $\ldots$ | $\ldots$ |
| 39 | 30.96 | 580.8 | 2.261 | 0.4071 | 0.2577 | 32.20 | 4500 | 7400 | 3300 |  |  |
| 40 | 30.96 | 688.0 | 1.892 | 0.4071 | $>0.1592$ |  | < 8600 | <19200 | $<10100$ | $\cdots$ | .. |
| 41 | 30.96 | 688.0 | 1.877 | 0.4071 | 0.1803 | 52.09 | 8300 | 16200 | 8600 | $\ldots$ |  |
| 42 | 30.96 | 1004.0 | 2.200 | 0.4071 | 0.1735 |  | 13900 | 33000 | 15000 | ... |  |
| 43 | 30.96 | 718.4 | 2.153 | 0.4071 | 0.2913 | 28.47 | 8500 | 11900 | 5500 | .. | . |
| 44 | 30.96 | 917.6 | 3.206 | 0.4071 | 0.1942 | 42.73 | 4800 | 14700 | 4600 |  |  |
| 45 | 30.96 | 524.6 | 2.019 | 0.4071 | 0.3441 | 24.12 | 5800 | 6300 | 3100 | ... |  |
| 48 | 30.96 | 481.7 | 2.021 | 0.4071 | 0.3495 | 23.76 | 4500 | 4900 | 2400 | ... | $\cdots$ |
| 49 | 30.96 | 638.9 | 1.942 | 0.4071 | 0.2875 | 28.87 | 9100 | 11300 | 5800 |  |  |
| 50 | 30.96 | 718.6 | 2.140 | 0.4071 | 0.2121 | 39.14 | 9800 | 18200 | 8500 | ... |  |
| 51 | 30.96 | 609.3 | 2.174 | 0.4071 | 0.2768 | 29.98 | 8700 | 12700 | 5800 |  |  |
| 114 | 30.98 | 715.5 | 2.762 | 0.4071 | 0.2228 | 37.34 | 5000 | 12700 | 4600 | $\ldots$ | $\ldots$ |
| 115 | 30.98 | 716.6 | 2.774 | 0.4071 | 0.2165 | 38.41 | 5000 | 12700 | 4600 |  | $\ldots$ |
| 116 | 30.98 | 716.5 | 2.770 | 0.4071 | 0.2232 | 37.25 | 5000 | 12700 | 4600 | $\ldots$ |  |
| 117 | 30.98 | 716.6 | 2.767 | 0.4071 | 0.2067 | 40.23 | 5100 | 13800 | 5000 |  |  |
| 337 | 32.04 | 460.2 | 2.874 | 0.4070 | 0.7358 | 11.28 | 7900 | 5700 | 2000 | $\ldots$ |  |
| 342 | 32.04 | 317.0 | 2.890 | 0.4070 | 1.7077 | 4.861 | 5100 | 1500 | 500 | ... | $\ldots$ |
| 344 | 32.04 | 600.1 | 2.891 | 0.4070 | 0.4868 | 17.07 | 9800 | 11000 | 3800 |  |  |
| 147 | 32.09 | 843.0 | 2.118 | 0.4068 | 0.0928 | 98.84 | 9200 | 47000 | 22200 | $\ldots$ |  |
| 151 | 32.09 | 264.1 | 1.155 | 0.4068 | 0.3083 | 26.94 | 5900 | 4200 | 3600 |  | 0.300 |
| 157 | 32.09 | 613.3 | 2.301 | 0.4068 | 0.2500 | 33.23 | 6500 | 11700 | 5100 |  | 0.566 |
| 166 | 32.09 | 215.1 | 1.167 | 0.4068 | 0.3779 | 21.99 | 5500 | 3200 | 2700 |  | 0.243 |
| 171 | 32.09 | 1036.0 | 4.616 | 0.4068 | 0.4976 | 16.69 | 5000 | 8800 | 1900 | . | 0.391 |
| 172 | 32.09 | 315.1 | 1.157 | 0.4068 | 0.1511 | 55.03 | 7700 | 11200 | 9700 |  | 0.481 |
| 175 | 32.09 | 740.9 | 2.343 | 0.4068 | 0.1634 | 50.88 | 7700 | 22700 | 9700 | $\ldots$ | ... |
| 176 | 32.09 | 488.7 | 2.315 | 0.4068 | 0.4161 | 19.96 | 5400 | 6000 | 2600 |  |  |
| 180 | 32.09 | 366.2 | 1.158 | 0.4068 | 0.1009 | 82.55 | 8900 | 19700 | 17000 | 0.08520 | ${ }^{0.683}$ |
| 181 | 32.09 | 871.7 | 2.296 | 0.4068 | 0.1049 | 77.54 | 9300 | 46200 | 20100 | 0.07858 | 1.068 |
| 182 | 32.09 | 1312.0 | 4.629 | 0.4068 | 0.3176 | 26.14 | 6000 | 17600 | 3800 | ... | 0.532 |

In Tests 337 through 407 a vent mechanism prevented piston from cscillating.
Closest piston approach measured with a lead gage. Maximum temperatures and pressures calculated from these assuming isentropic compression and no chemical reaction.
Experimental information not available.
Average value.


## Effect of maximum apparent temperature upon nitric oxide ratio

was established from a series of tests made with the piston weighing 3.29 pounds. Measurements were made upon mixtures which contained equal mole fractions of nitrogen and oxygen with 0.8 mole fraction helium and mixtures which contained 0.2 mole fraction nitric oxide with 0.8 mole fraction helium. The residual nitric oxide ratio resulting from the nitric oxide-helium sample decreased with an increase in the maximum apparent temperature (Figure 8). On the other hand, the residual nitric oxide ratio in the mixture of nitrogen and oxyger increased with an increase
in the maximum apparent temperature. Within the uncertainty of measurement, these two mixtures yielded the same residual composition at temperatures of the order of $10,000^{\circ}$ R. The residual composition corresponding to a nitric oxide ratio of $1.8 \times 10^{-3}$ was not the equilibrium composition, but the composition remaining after the sample was subjected to the variable cooling rate associated with these measurements. The maximum cooling rate attained was approximately $2^{\circ}$ R. per microsecond.

Figures 9 and 10 show the effect of rate of temperature

| Test No. | Initial Composition, ${ }^{\circ}$ Mole Fraction |  |  |  | Final Composition ${ }^{\text {b }}$ |  |  | Initial composition, ${ }^{\text {a }}$ Mole Fraction |  |  |  | Final Composition ${ }^{\text {b }}$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  | Nitric oxide, mole fraction | $\begin{gathered} \begin{array}{c} \text { Nitric Oxide } \\ \text { ratio } \\ \left(n_{\mathrm{NO}}\right)^{2} \end{array} \\ n n \end{gathered}$ |  |  |  |  |  | Nitric oxide, mole | Nitric oxide ratio $n_{\mathrm{YO}^{2}}{ }^{-2}$ |
|  | Nitrogen | Oxygen | Nitric oxide | Helium |  |  |  | Nitrogen | Oxygen | Nitric oxide | Helium | fraction | $\frac{n_{N_{2}} n_{0}}{}$ |
| 133 | 0.8250 | 0.1750 |  |  | 0.0061 | $0.2615 \times 10^{-3}$ | 158 | 0.4011 | 0.1993 |  | 0.3996 |  |  |
| 136 | 0.4664 | 0.1734 |  | 0.3602 | 0.0221 | 6.458 | 36 | 0.4626 | 0.5303 |  | 0.0044 | 0.0327 |  |
| 391 | 0.1002 | 0.0999 |  | 0.7999 | 0.0029 | 0.8953 | 38 | 0.1250 | 0.1248 |  | 0.7503 | 0.0102 | 7.320 |
| 392 | 0.1002 | 0.0999 |  | 0.7999 | 0.0032 | 1.0433 | 39 | 0.4349 | 0.4373 |  | 0.1278 | 0.0037 | 0.0723 |
| 394 | 0.1002 | 0.0999 |  | 0.7999 | 0.0019 | 0.3520 | 40 | 0.2435 | 0.2438 |  | 0.5126 | 0.0198 | 7.175 |
| 395 | 0.1002 | 0.0999 |  | 0.7999 | 0.0000 | 0.0000 | 41 | 0.2425 | 0.2437 |  | 0.5138 | 0.0190 | 6.629 |
| 397 | 0.1002 | 0.0999 |  | 0.7999 | 0.0042 | 1.8585 | 42 | 0.1251 | 0.1248 |  | 0.7501 | 0.0138 | 13.67 |
| 399 | 0.1328 | 0.0673 |  | 0.8000 | 0.0034 | 1.3666 | 43 | 0.1849 | 0.1839 |  | 0.6312 | 0.0129 | 5.274 |
| 401 | 0.1328 | 0.0673 |  | 0.8000 | 0.0043 | 2.129 | 44 | 0.4444 | 0.4513 |  | 0.1043 | 0.0120 | 0.7413 |
| 402 | 0.1328 | 0.0673 |  | 0.8000 | 0.0038 | 1.9696 | 45 | 0.4760 |  | 0.1272 | 0.3968 | 0.0457 | 99.44 |
| 403 | 0.1328 | 0.0673 |  | 0.8000 | 0.0033 | 1.3171 | 48 | 0.6530 |  | 0.1644 | 0.1826 | 0.0369 | 29.88 |
| 405 |  |  | 0.2007 | 0.7992 | 0.0138 | 21.78 | 49 | 0.1794 |  | 0.1818 | 0.6388 | 0.0119 | 6.406 |
| 406 |  |  | 0.2007 | 0.7992 | 0.0452 | 338.15 | 50 | 0.1839 | 0.1845 |  | 0.6316 | 0.0050 | 0.7567 |
| 407 |  |  | 0.2007 | 0.7992 | 0.0146 | 24.45 | 51 | 0.1846 | 0.1843 |  | 0.6312 | 0.0109 | 3.744 |
| 408 |  |  | 0.2007 | 0.7992 | 0.0108 | 13.02 | 114 | 0.4251 | 0.4246 |  | 0.1503 |  |  |
| 409 |  |  | 0.2007 | 0.7992 | 0.0064 | 4.379 | 115 | 0.4258 | 0.4248 |  | 0.1494 |  |  |
| 154 | 0.3978 | 0.2008 |  | 0.4013 | 0.0147 | 2.865 | 116 | 0.4252 | 0.4251 |  | 0.1497 |  |  |
| 162 | 0.3994 | 0.2001 |  | 0.4004 | 0.0151 | 3.041 | 117 | 0.4253 | 0.4253 |  | 0.1494 |  |  |
| 163 | 0.4008 | 0.2004 |  | 0.3988 | 0.0002 | 0.0005 | 337 | 0.1009 | 0.0974 |  | 0.8016 | 0.0019 | 0.3579 |
| 178 | 0.3996 | 0.2001 |  | 0.4003 | 0.0146 | 2.8085 | 342 | 0.1016 | 0.1010 |  | 0.7974 | 0.0000 | 0.0000 |
| 179 | 0.3996 | 0.1999 |  | 0.4005 | 0.0156 | 3.242 | 344 | 0.1001 | 0.1019 |  | 0.7979 | 0.0044 | 1.960 |
| 184 | 0.4039 | 0.2032 | . . $\cdot$ | 0.3929 | 0.0002 | 0.0005 | 147 | 0.4727 | 0.1672 |  | 0.3601 | 0.0116 | 1.7785 |
| 185 | 0.4030 | 0.1966 |  | 0.4003 | 0.0041 | 0.2170 | 151 | 0.3955 | 0.1980 |  | 0.4065 | 0.0008 | 0.0083 |
| 310 | . . . | . . . | 0.1995 | 0.8005 | 0.0203 | 51.12 | 157 | 0.3995 | 0.2007 |  | 0.3997 | 0.0108 | 1.521 |
| 313 |  |  | 0.1997 | 0.8003 | 0.0090 | 8.820 | 166 | 0.3996 | 0.2001 |  | 0.4003 | 0.0002 | 0.0004 |
| 315 |  |  | 0.1999 | 0.8001 | 0.0042 | 1.812 | 171 | 0.4005 | 0.2002 |  | 0.3992 | 0.0045 | 0.2881 |
| 317 | 0.1010 | 0.1034 |  | 0.7956 | 0.0000 | 0.0000 | 172 | 0.3994 | 0.2005 |  | 0.4001 | 0.0114 | 1.693 |
| 319 | 0.0961 | 0.0957 |  | 0.8082 | 0.0029 | 0.9657 | 175 | 0.4008 | 0.1994 |  | 0.3998 | 0.0129 | 2.194 |
| 321 | 0.1012 | 0.1012 |  | 0.7976 | 0.0042 | 1.761 | 176 | 0.3995 | 0.2005 | . . | 0.4001 | 0.0020 | 0.0529 |
| 146 | 0.4563 | 0.1718 |  | 0.3718 | 0.0116 | 1.796 | 180 | 0.3996 | 0.2001 |  | 0.4004 | 0.0122 | 1.951 |
| 149 | 0.4000 | 0.1987 |  | 0.4013 |  |  | 181 | 0.3995 | 0.1994 | . . | 0.4010 | 0.0132 | 2.287 |
| 150 | 0.3983 | 0.1984 |  | 0.4032 | 0.0108 | 1.5485 | 182 | 0.3870 | 0.2123 | . . | 0.4006 | 0.0093 | 1.0945 |
| 152 | 0.4003 | 0.1998 |  | 0.3999 | 0.0102 | 1.3715 |  |  |  |  |  |  |  |

${ }^{a}$ Initial compositions for Tests 36 through 51 were obtained by mass spectrographic analysis. For the other tests these data were computed from pressure and temperature measurements obtained $b$ during addition of the sample gases.
Final compositions for Tests 36 through 136 were obtained by mass spectrographic analysis. For the remaining tests final
compositions were obtained by a titrimetric method. The amount of various oxides of nitrogen in the final sample was converted to mole fraction of nitric oxide.
${ }^{c}$ Average of duplicate samples when available. Average deviation of 31 duplicate samples 0.0003 mole fraction nitric oxide and 0.0006 nitric oxide ratio.
change upon the nitric oxide ratio for initial sample pressures of 1.16 and 4.62 p.s.i., respectively. Values of the maximum apparent temperature for each point are included. As expected, there was a marked increase in the nitric oxide ratio with an increase in rate of temperature change. This increase in rate of temperature change, however, was accompanied by a corresponding increase in maximum apparent temperature, which accounts for the disparity between the data for the light and heavy pistons.

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Figure 7. Possible influence of oxygen upon the formation of nitric oxide


Figure 8. Residual nitric oxide ratio as a function of maximum apparent temperature

## NOMENCLATURE

| $K=$ | equilibrium constant |
| ---: | :--- |
| $k_{1}, k_{2}=$ | rate constants |
| $\left(\mathrm{N}_{2}\right),\left(\mathrm{O}_{2}\right),(\mathrm{NO})=$ | concentrations, lb.-mole/cu. ft. |
| $n_{k}=$ | mole fraction of component $k$ |
| $\underline{Q}=$ | total thermal transfer to face of piston |
|  | and bottom of compression chamber, |
| $T=$ | B.t.u. |
| $\Delta \underline{E}=$ | maximuture, ${ }^{\circ} \mathrm{R}$. |
|  | from piston to to sample energy transfer |
|  | to maximum compression, B.t.u. |


rate of temperature change, ${ }^{\circ} \mathrm{r}$. per microsec.
Figure 9. Sample pressure of 1.16 p.s.i.


Figure 10. Sample pressure of 4.62 p.s.i.

Subscripts

$$
\begin{aligned}
e & =\text { condition corresponding to equilibrium } \\
\max & =\text { maximum }
\end{aligned}
$$

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